



Decomposition Methods in MDO

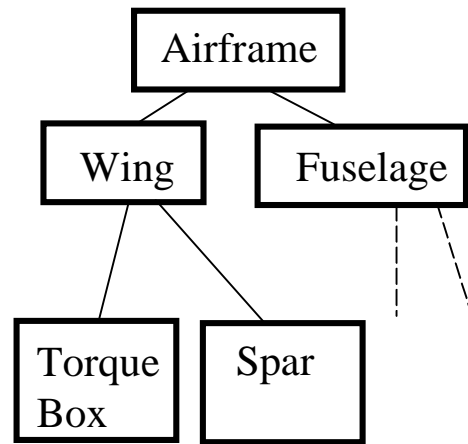


Decomposition Methods in Design

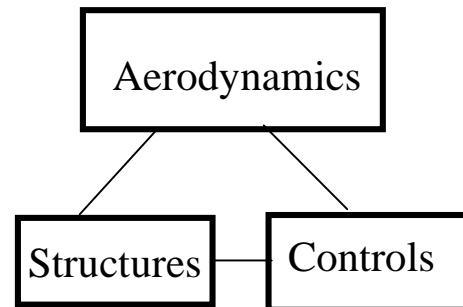
- Decomposition strategies have been used in the design of multidisciplinary systems
 - hierarchic decomposition
 - non-hierarchic decomposition
 - hybrid decomposition
- Elements of decomposition
 - partitioning
 - co-ordination



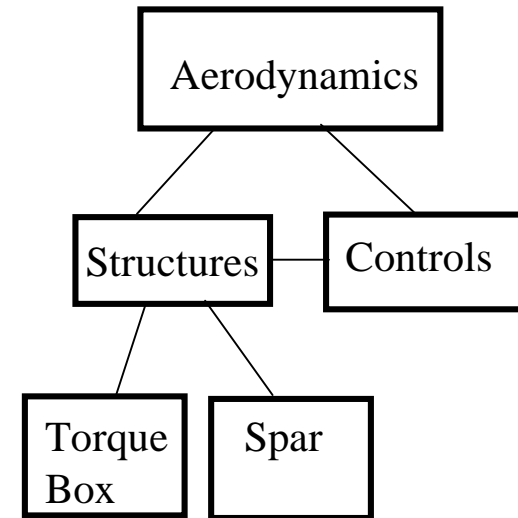
Decomposition Methods in Design



Hierarchic



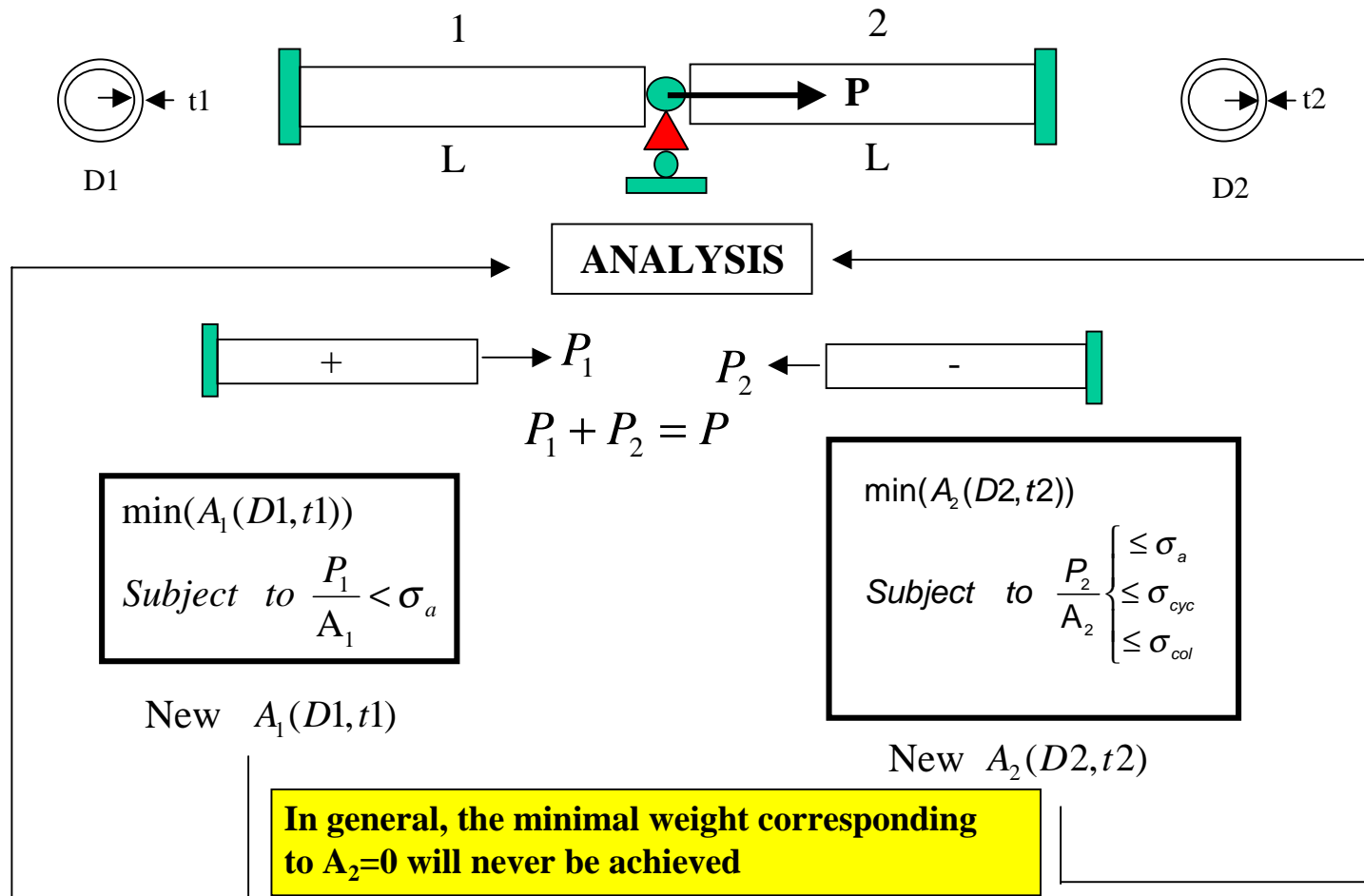
Non-Hierarchic



Hybrid

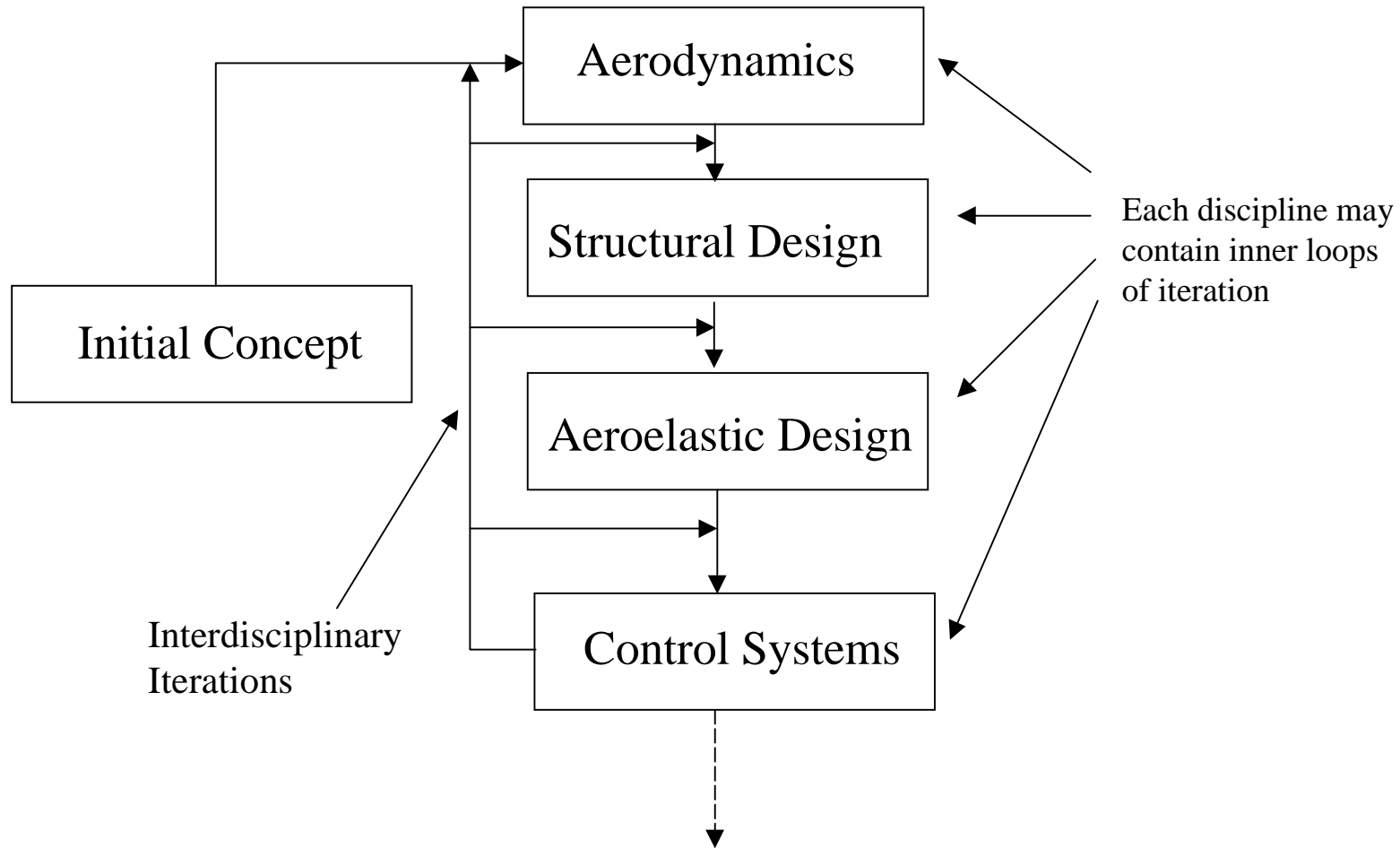


Partitioned Design W/O Coordination





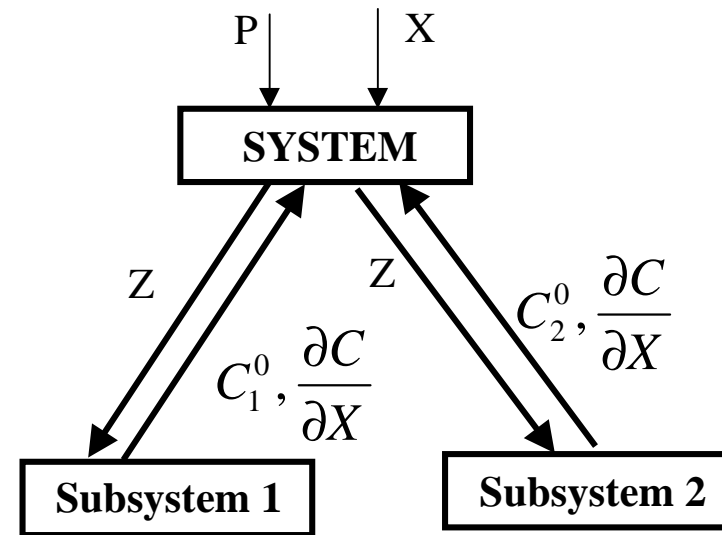
Sequential Approach





Hierarchic Decomposition Based Design

- System level
 - P - problem parameters
 - X - design variables
 - Q - behavior variables
 - Analysis $F(X,P,Q)=0$
 - Z passes as input to subsystem $Z=f(X,Q)$
- Subsystem level
 - x - local design variables
 - q - local behavior variables
 - analysis $f(Z,q,x)=0$





Hierarchic Decomposition Based Design

- Subsystem level optimization
 - $g=g(q(Z(X),x) \leq 0$ and $h=h(Z(X),x)=0$ are the inequality and equality constraints respectively
 - C is a cumulative constraint representation of all inequality constraints

- Optimization problem statement

Minimize C to obtain C_i^0
 x

Subject to $h = 0$

- Pass back $C_1^0, \frac{\partial C}{\partial X}$ and $C_2^0, \frac{\partial C}{\partial X}$ to the system level



Hierarchic Decomposition Based Design

- System level optimization
 - F is system objective $F=F(Q(X))$
 - G are system constraints $G=G(Q(X))$
- Mathematical problem statement

$$\underset{X}{\text{Minimize}} F(X)$$

$$\text{Subject to } G_j \leq 0; C_i \leq 0 \text{ for all } i \text{ subsystems}$$

- Here C_i are obtained as a linear extrapolation

$$C_i = C_i^0 + \frac{dC}{dX} \Delta X \quad \frac{dC}{dX} = \frac{\partial C}{\partial Z} \frac{\partial Z}{\partial X} \leftarrow \text{System sensitivity}$$

↑
Subsystem optimal sensitivity



Hierarchic Decomposition Based Design

- Errors in extrapolation due to active constraint switching
- Data management gets involved in realistic, large-scale MDO problems
- Handling of equality constraints is required as these provide system to subsystem coordination - proves to be problematic!



Non-Hierarchic Decomposition Based Design - CSSO

- Concurrent Subspace Optimization - Overview

Minimize $f(x^k)$

Subject to: $C^p \leq C^{po} [s^p (1 - r_k^p) + (1 - s^p) t_k^p]$ $p = 1, nss$

$$x_L^k \leq x^k \leq x_U^k$$

- C^p is a measure of all constraints in subspace p, super and subscript p and k denote the influence of subspace p on subspace k, r's are the responsibility coefficients, t's are the trade-off coefficients, and s are the switch parameters



CSSO Overview

- Optimum in each subspace is a function of r and t coefficients, and a second level problem needs to be solved - COP

$$\begin{aligned} \text{Minimize } F &= f^o + \sum_p \sum_k \frac{df}{dr_k} \Delta r_k^p + \sum_p \sum_k \frac{df}{dt_k} \Delta t_k^p \\ \text{Subject to: } \sum_k r_k^p &= 1 \quad p, k = 1, nss \\ \sum_k t_k^p &= 0 \quad 0 \leq r_k^p \leq 1 \\ r_{kL}^p &\leq r_k^p \leq r_{kU}^p \quad t_{kL}^p \leq t_k^p \leq t_{kU}^p \end{aligned}$$

- COP yields a new set of r's and t's to be used in next round of SSO's

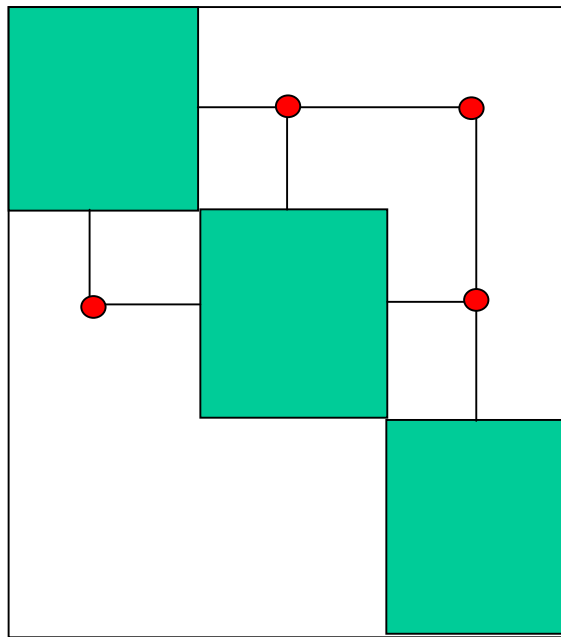


Deficiencies in CSSO

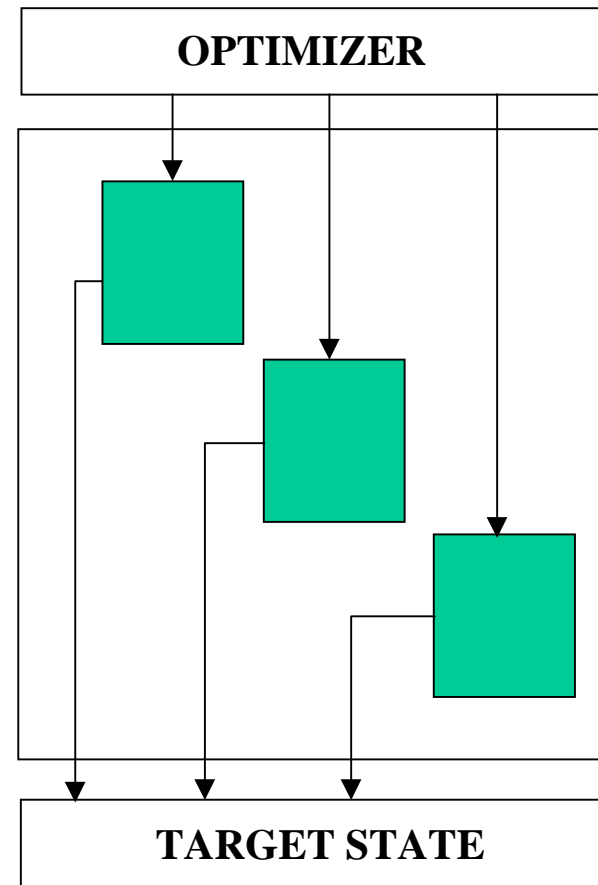
- Coupling in CSSO is resolved through the use of linear or higher-order approximations. These require move limits to be placed on the design variables.
- The formulation of the coordination problem is based on an optimal sensitivity analysis procedure that cannot be regarded as robust.
- The use of heuristics in solving the coordination problem have proven to be of limited benefit, often introducing cycling and convergence problems.
- Sensitivity information is unavailable when dealing with discrete and integer design variables.
- Solution of GSE for computing global sensitivity information is problematic in terms of the required computational effort, and numerical problems such as singularity and ill-conditioning.



Collaborative Optimization



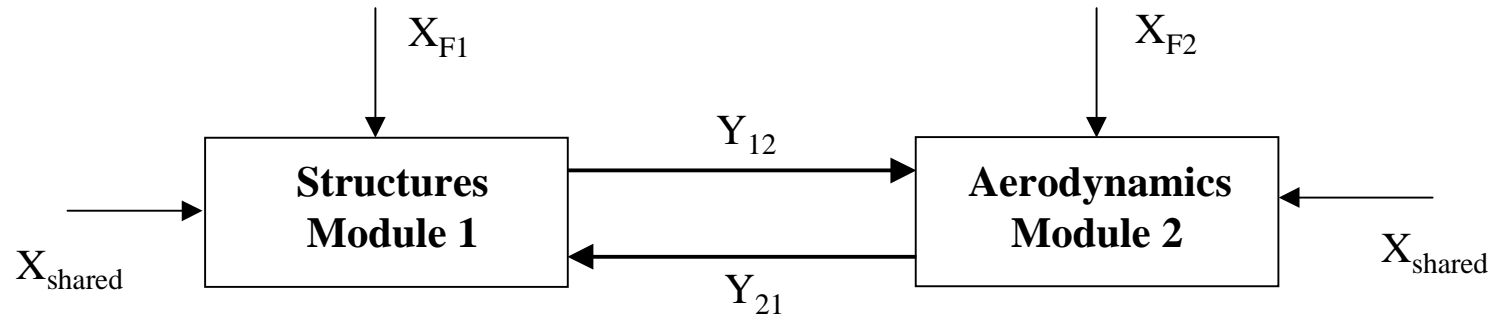
Coupled



Uncoupled



Collaborative Optimization - Example



X_{F1} - cross sectional dimensions of stringers and skins

X_M - Y_{21} and X_{shared}

$g(X)$ - constraints on stress and displacements

X_{F2} - airfoil leading edge radius and camber

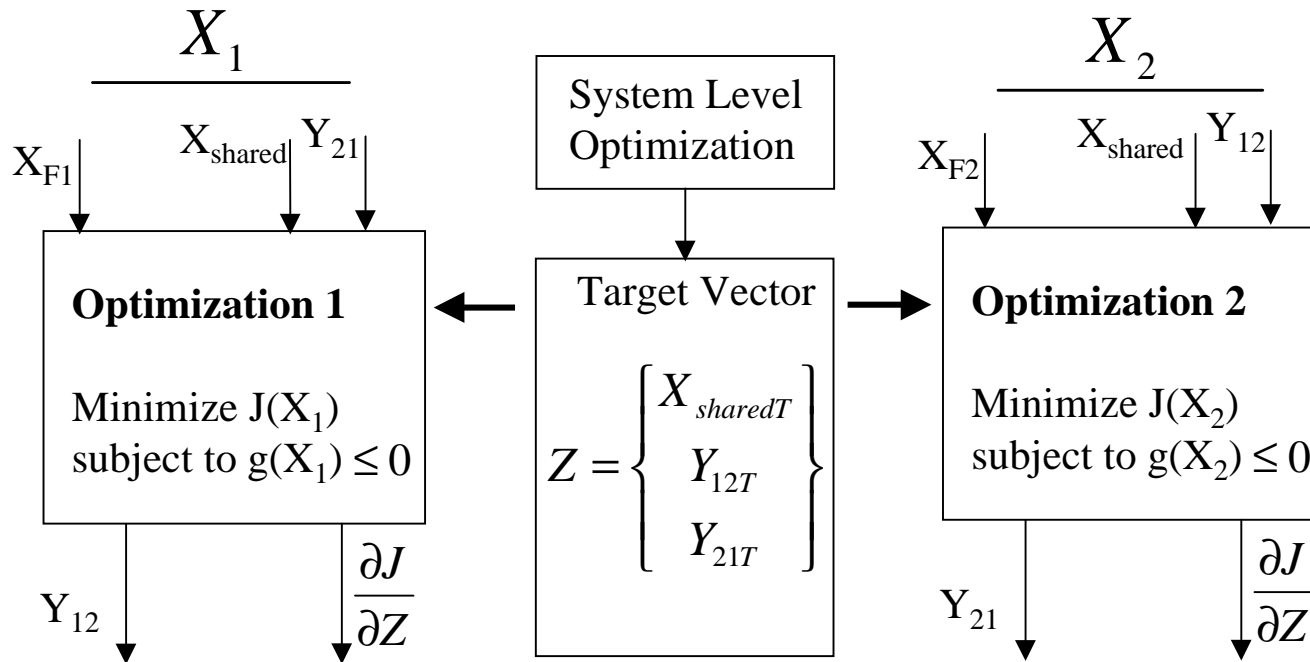
X_M - Y_{12} and X_{shared}

$g(X)$ - constraints on pressure distribution gradients

X_{shared} - wing taper, wing sweep, wing aspect ratio



Collaborative Optimization - Procedure



$$J = \begin{Bmatrix} Y_{21} - Y_{21T} \\ X_{shared} - X_{sharedT} \\ Y_{12} - Y_{12T} \end{Bmatrix}^T \begin{Bmatrix} Y_{21} - Y_{21T} \\ X_{shared} - X_{sharedT} \\ Y_{12} - Y_{12T} \end{Bmatrix} \quad || \quad J = \begin{Bmatrix} Y_{12} - Y_{12T} \\ X_{shared} - X_{sharedT} \\ Y_{21} - Y_{21T} \end{Bmatrix}^T \begin{Bmatrix} Y_{12} - Y_{12T} \\ X_{shared} - X_{sharedT} \\ Y_{21} - Y_{21T} \end{Bmatrix}$$



Collaborative Optimization System Level Optimization

- Find Z (system level variables shared by different modules) to minimize $F(Z)$ (system level objective)
- Satisfy $J_i=0$ for all modules I
 - here J_i is obtained from extrapolation based on optimal problem parameter sensitivity from each module

$$J_i = (J_i)_{old} + \left\{ \frac{\partial J_i}{\partial Z} \right\}^T \{ \Delta Z \}$$



Collaborative Optimization Summary

- Each module optimized separately - can be done concurrently
- System level coordination optimization is generally with small number of variables
- Each discipline is allowed to function autonomously
- There is no explicit system analysis - instead we have a situation akin to simultaneous analysis and design